WORKSHOP PROBLEM 3

Load Deflection of a 3-Rod Structure



Objectives:

- Demonstrate the use of geometric nonlinear analysis.
- Demonstrate the use of arc length increments.
- Create an animation of the deformation of the structure.
- Create an XY Plot of Load Factor vs. Displacement.

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Model Description:

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For the structure below:



Add Case Control commands and Bulk Data Entries to:

- 1. Perform a geometric nonlinear analysis.
- 2. Compute the load-deflection behavior of the three-rod structure shown above.

Suggested Exercise Steps:

- Modify the existing MSC/NASTRAN input file by adding the appropriate loading conditions and nonlinear static analysis control parameters.
- For Case Control, insert nonlinear static analysis parameter selection (NLPARM).
- For Bulk Data, insert all relevant nonlinear static analysis parameters (NLPARM).
- Prepare the model for a nonlinear static analysis.
 - PARAM, LGDISP, 1
- Insert parameters for arc-length methods (NLPCI).
- Generate an input file and submit it to the MSC/NASTRAN solver for a nonlinear static analysis.
- Review the results.

Input File for Modification:

prob3.dat

ID NAS103, WORKSHOP 3 **TIME 10** SOL 106 CEND TITLE=GEOMETRIC NONLINEAR PROBLEM SUBTITLE=REF: POWELL AND SIMONS, IJNME, 17:1455-1467, 1981 DISP=ALL SPCF=ALL OLOAD=ALL SPC=12 LOAD=10 **BEGIN BULK \$ GEOMETRY** GRID,1,,0.0,0. GRID,2,,5.,0. GRID,3,,10.,3. GRID,4,,10.,8. GRDSET,,,,,,3456 **\$ CONNECTIVITY** CONROD,1,1,2,10,.5 CONROD,2,2,3,11,1. CONROD,3,3,4,11,1. **\$ PROPERTIES** MAT1,10,1.+6 MAT1,11,3.+6 **\$ CONSTRAINTS** SPC1,12,1,3 SPC1,12,2,1,2 SPC1,12,12,4 \$ LOADING FORCE,10,1,,4.+5,1.,0.,0. **\$ PARAMETERS** PARAM,POST,0 **\$ SOLUTION STRATEGY** ENDDATA

Exercise Procedure:

1. Users who are not utilitizing MSC/PATRAN for generating an input file should go to Step 11, otherwise, proceed to step 2.

2. Create a new database called **prob3.db**.

File/New...

New Database Name

prob3

OK

In the New Model Preference form set the following:

Tolerance

Analysis Code:

Analysis Type:

OK

Default	
MSC/NASTRAN	
Structural	

3. Those who do not wish to set up the model themselves may want to play the session file, **prob3.ses**. If you choose to build the model yourself, proceed to step 4.

File/Session/Play...

Session File List

prob3.ses

Apply

The model has now been created. Skip to Step 10.

Whenever possible click **Auto Execute** (turn off).

4. Create the baseline NASTRAN finite element model.

♦ Finite Elements

Action:	Create
Object:	Node
Method:	Edit
	□ Associate with Geometry
Node ID List	1

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For clarity, increase the node size and turn on the entity labels by using the following toolbar icons:



5. Create the rod elements for the model.

Finite Elements

Action:	Create
Object:	Element
Method:	Edit
Shape:	Bar
<i>Node 1</i> =	Node 1
Node $2 =$	Node 2

Apply

Repeat the above proceedure to create the other two elements.



6. Create the material properties for the beams.

♦ Materials

Action:

Object:

Method:

Material Name:

Input Properties...

Elastic Modulus =

Apply

Cancel

Material Name:

Input Properties...

Elastic Modulus =



7. Define the rod properties.

Properties

Action:

Create	

Isotropic

Manual Input

mat_1

1.E6

mat_2

3.E6

Create

Object:	1D
Method:	Rod
Property Set Name	rod_1
Input Properties	
Material Name	m:mat_1
Area:	0.5
ОК	

In order to pick the beam elements, click on the following entity select icon from the small menu window:



Beam Element

Select Members

Elm 1

Add
Apply

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Create the second property set.

Property Set Name

rod_2

mat_2

Elm 2, 3

1.0

Input Properties...

Material Name

Area:

OK

Select Members

Add	
Apply	

8. Create the load for the model.

♦ Loads/BCs

Action:

Create

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Object	Form
Objeci.	Force
Method:	Nodal
New Set Name	load_1
Input Data	
<i>Force</i> < <i>F1 F2 F3</i> >	<4.E5, 0, 0>
ОК	
Select Application Region	
Geometry Filter	● FEM
Select Nodes	Node 1
Add	
OK	

9. Create the three boundary constraints for the model.

♦ Loads/BCs Action: Create Object: Displacement Nodal Method: New Set Name constraint_1 Input Data... <0, , > *Translation* < *T1 T2 T3* >: OK Select Application Region... Select Nodes: Node 3 Add OK Apply

constraint_2

New Set Name

Apply

Input Data	
Translation < T1 T2 T3 >:	< ,0, >
ОК	
Select Application Region	
Select Nodes:	Node 1, 2
Add	
ОК	
Apply	
New Set Name:	constraint_3
Input Data	
Translation < T1 T2 T3 >:	< 0, 0, >
ОК	
Select Application Region	
Select Nodes:	Node 4
Add	
ОК	
Apply	

10. Now generate the analysis input file.

Click on the **Analysis** radio button on the Top Menu Bar and set up the analysis as follows:



Action:

Object:

Method:

Job Name

Solution Type...

Analyze
Entire Model
Analysis Deck
prob3



OK	
Apply	

An input file called **prob3.bdf** will be generated. This process of translating your model into an input file is called the Forward Translation. The Forward Translation is complete when the Heartbeat turns green. MSC/PATRAN users should now proceed to **Step 12**.



Generating an input file for MSC/NASTRAN Users:

11. MSC/NASTRAN users can generate an input file using the data from the Model Description. The result should be similar to the output below (**prob3.dat**):

ASSIGN OUTPUT2 = 'prob3.op2', UNIT = 12 ID NAS103, WORKSHOP 3 SOLUTION **TIME 10** SOL 106 CEND TITLE=GEOMETRIC NONLINEAR PROBLEM SUBTITLE=REF: POWELL AND SIMONS, IJNME, 17:1455-1467, 1981 DISP=ALL SPCF=ALL OLOAD=ALL SPC=12 LOAD=10 SUBCASE 10 NLPARM=10 **BEGIN BULK \$ GEOMETRY** GRID,1,,0.0,0. GRID,2,,5.,0. GRID,3,,10.,3. GRID,4,,10.,8. GRDSET,,,,,,3456 **\$ CONNECTIVITY** CONROD,1,1,2,10,.5 CONROD,2,2,3,11,1. CONROD,3,3,4,11,1. **\$ PROPERTIES** MAT1,10,1.+6 MAT1,11,3.+6 **\$ CONSTRAINTS** SPC1,12,1,3 SPC1,12,2,1,2 SPC1,12,12,4

\$ LOADING FORCE,10,1,,4.+5,1.,0.,0. \$ PARAMETERS PARAM,POST,0 \$ SOLUTION STRATEGY NLPARM,10,40,,,,,YES NLPCI,10 PARAM,LGDISP,1 ENDDATA

Submit the input file for analysis:

- 12. Submit the input file to MSC/NASTRAN for analysis.
 - 12a. To submit the MSC/PATRAN **.bdf** file, find an available UNIX shell window. At the command prompt enter **nastran prob3.bdf scr=yes**. Monitor the analysis using the UNIX **ps** command.
 - 12b. To submit the MSC/NASTRAN .dat file, find an available UNIX shell window and at the command prompt enter nastran prob3.dat scr=yes. Monitor the analysis using the UNIX ps command.
- 13. When the analysis is completed, edit the **prob3.f06** file and search for the word **FATAL**. If no matches exist, search for the word **WARNING**. Determine whether existing WARNING messages indicate modeling errors.
- 13a. While still editing **prob3.f06**, search for the word:
- **DISPLACE** (spaces are necessary).

What is the x-displacement of **Node 1** at load step = 1.0?

T1 =

What is the x-displacement of **Node 2** at load step = 1.0?

T1 =

14. Compare the results obtained in the **.f06** file with the results on the following page:

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SUBCASE 1

LOAD STEP = 1.00000E+00

DISPLACEMENT VECTOR

POINT ID.	TYPE	Τ1	Т2	Т3	R1	R2	R3
1	G	1.510921E+01	0.0	0.0	0.0	0.0	0.0
2	G	1.110920E+01	0.0	0.0	0.0	0.0	0.0
3	G	0.0	-2.951657E-01	0.0	0.0	0.0	0.0
4	G	0.0	0.0	0.0	0.0	0.0	0.0



15. This ends the exercise for MSC/NASTRAN users. MSC/PATRAN users should proceed to the next step.

16. Proceed with the Reverse Translation process, that is, importing the **prob3.op2** results file into MSC/PATRAN. To do this, return to the **Analysis** form and proceed as follows:

♦ Analysis

Action:

Object:

Method:

Select Results File...

Selected Results File:

Read Output2
Result Entities
Translate

prob3.op2

OK	
Apply	

When the translation is complete and the Heartbeat turns green, bring up the **Results** form.

17. Create an animation that demonstrates the snap-through deformation.

Now we will generate the fringe plot of the model.

♦ Results

Action:

Object:

Create	
Fringe	

Now click on the Select Results icon.



Select Results

Make sure that the **View Subcases** icon is *off*. It should appear like the following.

ч.	_
- 13	_
- 1-4	
	_
	Ē

View Subcases

Select Result Case(s)

Select Fringe Result

Quantity:

Displacements, Translational

Next click on the Target Entities icon.



Target Entities

Target Entity:

Current Viewport

Note: This feature allows you to view fringe plots of specific elements of your choice.

Click on the **Display Attributes** icon.



Display Attributes

Style:

Display:

Discrete/Smooth Free Edges

For better visual quality of the fringe plot, change the width of the line.

Width:

(Select the third line from top.)

Note: The **Display Attributes** form allows you the ability to change the displayed graphics of fringe plots.

Now click on the **Plot Options** icon.



Coordinate Transformation:

Scale Factor

None	
1.0	

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Apply

The resulting fringe plot should display the displacement spectrum superimposed over the undeformed bar. The final fringe plot displaying the physical deformation of the model can be created as follows:

♦ Results

Action:

Object:

Create	
Deformation	

Now click on the Select Results icon.



Select Results

Select Result Case(s)

Select Fringe Result

Show As:

	Displacements.	Translational
--	----------------	---------------

Resultant

(Select the last case.)

Click on the **Display Attributes** icon.



Display Attributes

In order to see the deformation results accurately, set the Scale Interpretation to True Scale with a Scale Factor of 1.

Scale Interpretation

Scale Factor



■ Show Undeformed

Line Width:

(Select the third line from top.)

Now click on the **Plot Options** icon.



Coordinate Transformation:

None	
1.0	

Apply

Scale Factor

Now click on the **Animation Options** icon.



Animation Options

Animation Method:

Animation Graphics

Number of Frames:

Interpolation

Apply

To better fit the results on the screen, zoom out a couple times using the following toolbar icon:



Alternatively, use any number of the toolbar icons to better view the resulting fringe plot.

You can see the physical deformation of the model as well as the amount of deformation from the fringe.

Notice the loading on the middle section shifts from compression to tension as the lower section "snaps-through" the initial setup. This in an instance when nonlinear analysis is definitely required to obtain an accurate solution.

When you wish to stop the animation, click on the following button:

Stop Animation

When asked if you wish to clear the current animation, select Yes.

Yes

18. Create an XY plot of X-Displacement vs. Percent of Load.

Create an XY plot of Load Factor vs. Displacement.

Results

Action:	Create
Object:	Graph
Method:	Y vs X

Select all the Result Cases by highlighting them.

Select Result Case(s)	(Select all cases.)
<i>Y</i> :	Result
Select Y Result	Applied Loads, Translational
Quantity:	X Component
<i>X</i> :	Result
Select X Result	
Select X Result	Displacements, Translational
Quantity:	X Component
ОК	

Next click on the Target Entities icon.



Target Entities

Select Nodes

Nodes	
Node 1	

(Select node	at e	end of	beam.)
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Click on the	Display	Attributes	icon.
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Display Attributes

V Aris Label	Displacement
A AXIS LUDEI.	Displacement
X Axis Scale	Linear
X Axis Format	
Label Format:	Fixed
OK	
Show Y Axis Label	
Y Axis Label:	Applied Load
'Axis Scale	● Linear
Y Axis Format	
Label Format:	Fixed
	L

Now click on the **Plot Options** icon



Plot Options

Coordinate Transformation:

None	
1.0	

Scale Factor Apply

To change the title, do the following:

♦ XY Plot

Action:

Object:

Curve List

Title...

Modify

Curve

default_GraphResults Graph 0

Load Deflection of 3-Rod Structure

Curve Title Text

Load Factor vs. Displacement @ Node 1

Apply	
Cancel	

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The following XY plot should appear on your screen:



Notice the load factor makes a dramatic shift to negative values when the deformation is between 5 to 6 inches. This is caused by the snapthrough behavior of the model. Eventually, the load does reach its full (positive) value as the snap-through effects die out.

Again, behavior such as this can only be determined by a geometric nonlinear analysis. A linear static analysis would not reveal this behavior.

When done viewing, delete the XY plot by doing the following:



Action:

Object:

Post	
XYWindow	

Post/Unpost XY Windows:

(hold <*ctrl*> and click on **Results Graph** to deselect it.)

Apply

Quit MSC/PATRAN when you have completed this exercise.

MSC/PATRAN .bdf file: prob3.bdf

```
$ NASTRAN input file created by the MSC MSC/NASTRAN input file
$ translator (MSC/PATRAN Version 7.5) on January 15, 1998 at
$ 14:08:17.
ASSIGN OUTPUT2 = 'prob3.op2', UNIT = 12
$ Direct Text Input for File Management Section
$ Nonlinear Static Analysis, Database
SOL 106
TIME 600
$ Direct Text Input for Executive Control
CEND
SEALL = ALL
SUPER = ALL
TITLE = MSC/NASTRAN job created on 15-Jan-98 at 14:06:49
ECHO = NONE
MAXLINES = 999999999
$ Direct Text Input for Global Case Control Data
SUBCASE 1
$ Subcase name : Default
 SUBTITLE=Default
 NLPARM = 1
 SPC = 2
 LOAD = 2
 DISPLACEMENT(SORT1,REAL)=ALL
 SPCFORCES(SORT1,REAL)=ALL
 OLOAD(SORT1,REAL)=ALL
$ Direct Text Input for this Subcase
BEGIN BULK
PARAM POST -1
PARAM PATVER 3.
PARAM AUTOSPC YES
PARAM COUPMASS -1
PARAM K6ROT 100.
PARAM WTMASS 1.
PARAM LGDISP 1
PARAM, NOCOMPS, -1
PARAM PRTMAXIM YES
NLPARM 1
              40
                       AUTO 5
                                   25
                                        PW
                                              YES +
                                                        Α
          .001 1.-7
+
   Α
$ Direct Text Input for Bulk Data
NLPCI. 1
$ Elements and Element Properties for region : rod_1
```

PROD 1 .5 1 CROD 1 1 2 1 \$ Elements and Element Properties for region : rod_2 PROD 2 2 1. 2 2 2 3 CROD CROD 4 3 2 3 **\$** Referenced Material Records \$ Material Record : mat_1 \$ Description of Material : Date: 30-May-97 Time: 17:00:10 MAT1 1.+61 \$ Material Record : mat_2 \$ Description of Material : Date: 30-May-97 Time: 17:00:10 MAT1 2 3.+6 \$ Nodes of the Entire Model GRID 1 0. 0. 0. 2 5. 0. GRID 0. GRID 3 10. 3. 0. GRID 4 10. 8. 0. \$ Loads for Load Case : Default SPCADD 2 1 3 4 LOAD 2 1. 1. 1 \$ Displacement Constraints of Load Set : constraint_1 SPC1 1 1 3 \$ Displacement Constraints of Load Set : constraint_2 SPC1 3 2 1 2 \$ Displacement Constraints of Load Set : constraint_3 SPC1 4 12 4 \$ Nodal Forces of Load Set : load_1 1 0 FORCE 1 400000.1. 0. 0. **\$** Referenced Coordinate Frames ENDDATA b915b9bd