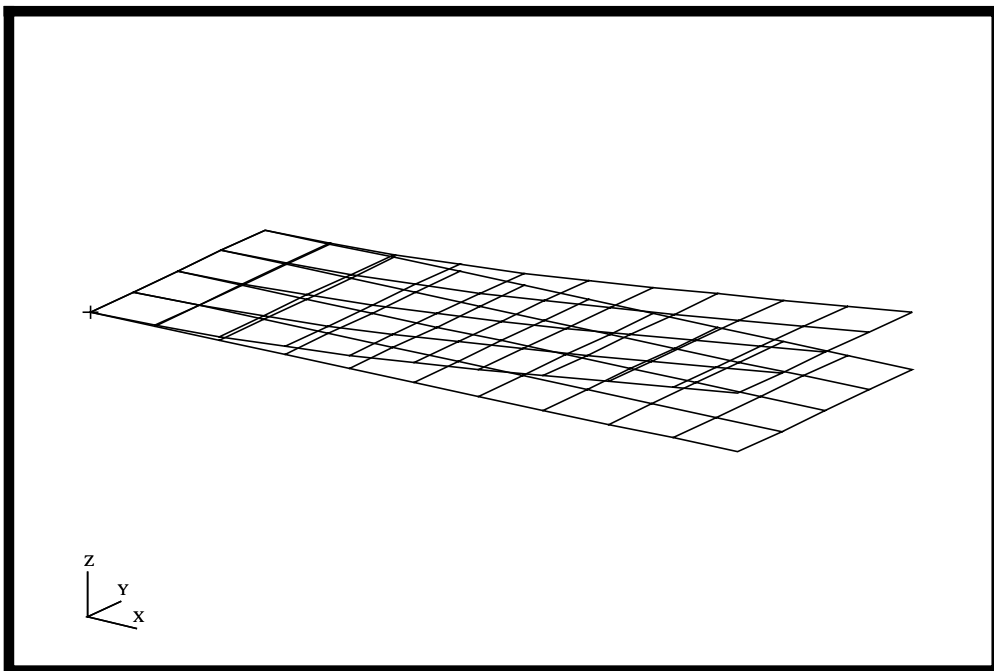


---

## WORKSHOP PROBLEM 7

---

# *Direct Transient Response with Base Excitation*



### Objectives

- Define a time-varying unit acceleration.
- Use the large mass method.
- Produce a MSC/NASTRAN input file from a dynamic math model, created in Workshop 1.
- Submit the file for analysis in MSC/NASTRAN.
- Compute nodal displacements for desired time domain.

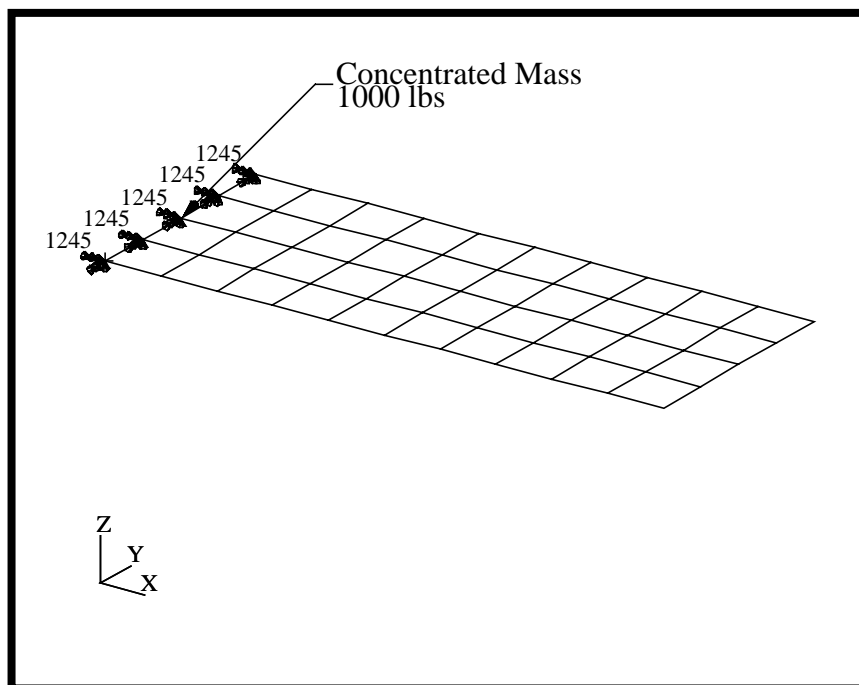


## Model Description:

Using the direct method, determine the transient response to a unit acceleration sine pulse of 250 Hz applied at the base in the z direction. A large mass of 1000 lb is applied to the base. Use a structural damping coefficient of  $g = 0.06$  and convert this damping to equivalent viscous damping at 250 Hz.

Below is a finite element representation of the flat plate. It also contains the loads and boundary constraints.

**Figure 7.1-**Loads and Boundary Conditions



---

## Suggested Exercise Steps

- Reference previously created dynamic math model, **plate.bdf**, by using the INCLUDE statement.
- Modify base constraints and release displacements in the Z-direction.
- Define the time-varying unit acceleration (TLOAD2 and DAREA).
- Create the large mass at the base (CMASS2 and RBE2).
- Specify the structural damping and convert this damping to equivalent viscous damping.
  - PARAM, G, 0.06
  - PARAM, W3, 1571
- Specify integration time steps (TSTEP).
- Prepare the model for a direct transient analysis (SOL 109).
- Request response in terms of nodal displacement, velocity and acceleration at Grids 11, 33 and 55.
- Generate an input file and submit it to the MSC/NASTRAN solver for normal modes analysis.
- Review the results, specifically the nodal displacements, velocities, and acceleration.







---

## Exercise Procedure:

1. Users who are not utilizing MSC/PATRAN for generating an input file should go to Step 11, otherwise, proceed to step 2.
2. Create a new database and named **prob7.db**

### File/New Database

*New Database Name*

**prob7**

**OK**

In the *New Model Preference* form set the following:

*Tolerance*

◆ **Default**

*Analysis code:*

**MSC/NASTRAN**

**OK**

3. Create the model by importing an existing MSC/NASTRAN input file, (**plate.bdf**).

### ◆ Analysis

*Action:*

**Read Input File**

*Object:*

**Model Data**

*Method:*

**Translate**

**Select Input File**

*Select Input File*

**plate.bdf**

**OK**

**Apply**

**OK**

4. Activate the entity labels by selecting the Show Labels icon on the tool-bar.



**Show Labels**



**Iso 3 View**



5. Create a time dependent load case for the transient response..

◆ **Load Cases**

<i>Action:</i>	<input type="text" value="Create"/>
<i>Load Case Name:</i>	<input type="text" value="transient_response"/>
<i>Load Case Type:</i>	<input type="text" value="Time Dependent"/>
<i>Assign/Prioritize loads/BCs</i> <i>(Highlight the following:)</i>	<input type="text" value="Displ_spc1.1"/>

6. Place the large mass at the base. To do this, it will be necessary to create a point element at Node 23.

◆ **Finite Elements**

<i>Action:</i>	<input type="text" value="Create"/>
<i>Object:</i>	<input type="text" value="Element"/>
<i>Method:</i>	<input type="text" value="Edit"/>
<i>Shape:</i>	<input type="text" value="Point"/>
<i>Node 1 =</i>	<input type="text" value="Node 23"/>

7. Next, define the scalar mass.

◆ **Properties**

<i>Action:</i>	<input type="text" value="Create"/>
<i>Dimension:</i>	<input type="text" value="0 D"/>
<i>Type:</i>	<input type="text" value="Mass"/>
<i>Property Set Name:</i>	<input type="text" value="scalar_mass"/>
<i>Option(s)</i>	<input type="text" value="Grounded"/>

---

**Input Properties...**

Mass

1000

DOF at Node 1

UZ

(Select **String (UZ)** from the button on the right.)

**OK**

Select Members:

Elm 41

(Change the select menu from **point** to **point element**, then select the **point element** which was previously created.)



Point element

**Add**

**Apply**

8. Create the RBE Mass, which will be connected to the remaining base points.

◆ **Finite Elements**

Action:

Create

Object:

MPC

Type:

RBE2

**Define Terms ...**

(Turn off **Auto Execute**.)

Auto Execute

◆ **Create Independent**

Node List:

Node 23

**Apply**

◆ **Create Dependent**

Node List:

Node 1, 12, 34, 45

*DOFs*

9. Create the time-dependent field for the unit acceleration.

◆ **Fields**

*Action:*

*Object:*

*Method:*

*Field Name:*

*Maximum Number of t:*

*PCL Expression  $f'(t)$ :*

*Start time:*

*End time:*

*Number of Points:*

Go back to the *Time/Frequency Scalar Table Data* window. Add the following to row 21:

	<b>Time(t)</b>	<b>Value</b>
<input type="text" value="21"/>	<input type="text" value="0.04"/>	<input type="text" value="0.0"/>
<input type="button" value="OK"/>		
<input type="button" value="Apply"/>		

---

9a. Now define the unit acceleration.

◆ **Loads/BCs**

*Action:*

**Create**

*Object:*

**Force**

*New Set Name:*

**unit\_acceleration**

**Input Data...**

*Spatial Dependence:*

**<0, 0, 2.588>**

*Time Dependence:*

**f:time\_dependent\_acceleration**

**OK**

**Select Application Region ...**

◆ **FEM**

**Node 23**

**Add**

**OK**

**Apply**

9b. Finally, modify the existing constraints by releasing DOF3.

◆ **Load/BCs**

*Action:*

**Modify**

*Object:*

**Displacement**

*Type:*

**Nodal**

*Select Set to Modify*

**spc1.1**

**Modify Data...**

*Translations < T1 T2 T3 >*

**< 0, 0, >**

*Rotations < R1 R2 R3 >*

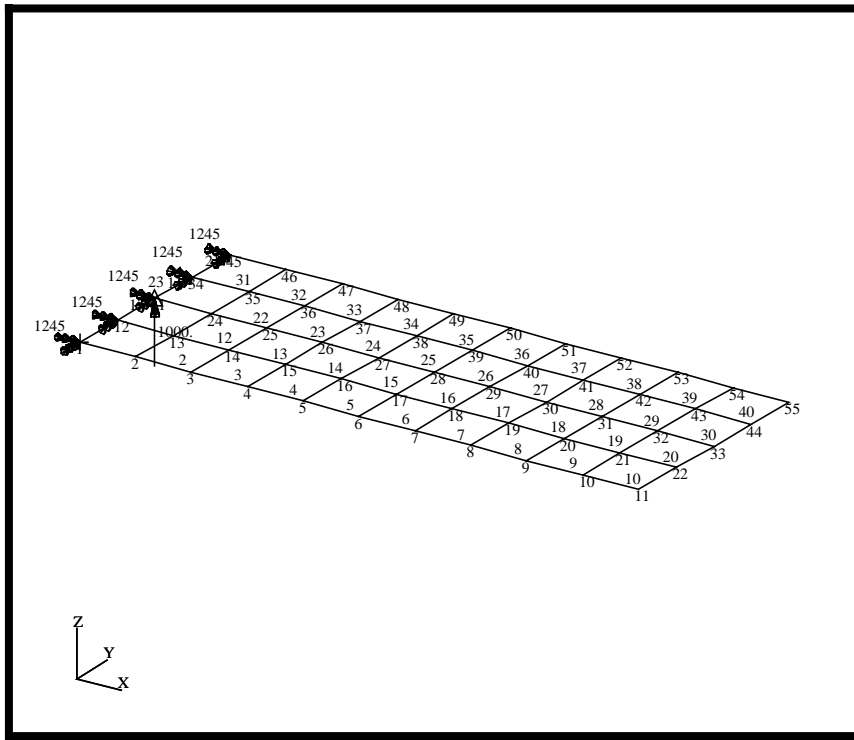
**< 0, 0, >**

**OK**

**Apply**

9c. The result should look like Figure 7.2

**Figure 7.2**



10. Generate the input file.

◆ **Analysis**

*Action:*

**Analyze**

*Object:*

**Entire Model**

*Method:*

**Analysis Deck**

*Job Name:*

**prob7**

**Solution Type...**

*Solution Type:*

◆ **TRANSIENT RESPONSE**

**Solution Parameters ...**

*Formulation*

**Direct**

*Mass Calculation*

**Coupled**

*Wt.-Mass Conversion*

**.00259**

---

*Struct. Dampening Coeff.* =

*W3, Damping Factor* =

*Available Subcases*

*Ending Time* =

*Number of Time Steps* =

under *Select Result Type*,  
highlight:

under *Output Requests*, highlight:

**SPCFORCES(SORT1,Real)=ALL FEM**

*Subcases Selected:*  
*(Click to deselect.)*

*Subcases for Solution*  
*Sequence: 109*  
*(Click to select.)*

An MSC/NASTRAN input file called **prob7.bdf** will be generated. This process of translating your model into an input file is called the Forward Translation. The Forward Translation is complete when the Heartbeat turns green. MSC/PATRAN Users should proceed to step 12.

---

## Generating an input file for MSC/NASTRAN Users:

MSC/NASTRAN users can generate an input file using the data previously stated. The result should be similar to the output below.

### 11. MSC/NASTRAN Input File: **prob7.dat**

```
ID SEMINAR, PROB7
SOL 109
TIME 30
CEND
TITLE = TRANSIENT RESPONSE WITH BASE EXCITATION
SUBTITLE = USING DIRECT TRANSIENT METHOD, NO REDUCTION
ECHO = UNSORTED
SPC = 200
SET 111 = 23, 33
DISPLACEMENT (SORT2) = 111
VELOCITY (SORT2) = 111
ACCELERATION (SORT2) = 111
SUBCASE 1
DLOAD = 500
TSTEP = 100
$
OUTPUT (XYPLOT)
XGRID=YES
YGRID=YES
XTITLE= TIME (SEC)
YTITLE= BASE ACCELERATION
XYPLOT ACCELERATION RESPONSE / 23 (T3)
YTITLE= BASE DISPLACEMENT
XYPLOT DISP RESPONSE / 23 (T3)
YTITLE= TIP CENTER DISPLACEMENT RESPONSE
XYPLOT DISP RESPONSE / 33 (T3)
$
BEGIN BULK
$
$ PLATE MODEL DESCRIBED IN NORMAL MODES EXAMPLE
$
INCLUDE 'plate.bdf'
PARAM, COUPMASS, 1
PARAM, WTMASS, 0.00259
$
$ SPECIFY STRUCTURAL DAMPING
$
PARAM, G, 0.06
```



```
PARAM, W3, 1571.
$
$ APPLY EDGE CONSTRAINTS
$
SPC1, 200, 12456, 1, 12, 23, 34, 45
$
$ PLACE BIG FOUNDATION MASS (BFM) AT BASE
$
CMASS2, 100, 1000., 23, 3
$
$ RBE MASS TO REMAINING BASE POINTS
$
RBE2, 101, 23, 3, 1, 12, 34, 45
$
$ APPLY LOADING TO FOUNDATION MASS
$
TLOAD2, 500, 600, , 0, 0.0, 0.004, 250., -90.
$
DAREA, 600, 23, 3, 2.588
$
$ SPECIFY INTEGRATION TIME STEPS
$
TSTEP, 100, 200, 2.0E-4, 1
$
ENDDATA
```

---

## Submitting the input file for analysis:

12. Submit the input file to MSC/NASTRAN for analysis.

12a. To submit the MSC/PATRAN **.bdf** file for analysis, find an available UNIX shell window. At the command prompt enter: **nastran prob7.bdf scr=yes**. Monitor the run using the UNIX **ps** command.

12b. To submit the MSC/NASTRAN **.dat** file for analysis, find an available UNIX shell window. At the command prompt enter: **nastran prob7 scr=yes**. Monitor the run using the UNIX **ps** command.

13. When the run is completed, use **plotps** utility to create a postscript file, **prob7.ps**, from the binary plot file **prob7.plt**. The displacement, velocity, and acceleration response plots for Grids 11, 33 and 55 are shown in figures 7.3 to 7.8.

14. When the run is completed, edit the **prob7.f06** file and search for the word **FATAL**. If no matches exist, search for the word **WARNING**. Determine whether existing **WARNING** messages indicate modeling errors.

15. While still editing **prob7.f06**, search for the word:

**D I S P L** (spaces are necessary)

Displacement at Grid 23

Time        T3

.0    = \_\_\_\_\_

.02   = \_\_\_\_\_

.04   = \_\_\_\_\_

Displacement at Grid 33

Time        T3

.0    = \_\_\_\_\_

.02   = \_\_\_\_\_

.04   = \_\_\_\_\_

**V E L O C** (spaces are necessary)

Velocity at Grid 23

Time        T3

.0    = \_\_\_\_\_

.02   = \_\_\_\_\_

.04   = \_\_\_\_\_

Velocity at Grid 33

Time        T3

.0    = \_\_\_\_\_

.02   = \_\_\_\_\_

.04   = \_\_\_\_\_

---

**A C C E L** (spaces are necessary)

Acceleration at Grid 23

Time        T3

.0 = \_\_\_\_\_

.02 = \_\_\_\_\_

.04 = \_\_\_\_\_

Acceleration at Grid 33

Time        T3

.0 = \_\_\_\_\_

.02 = \_\_\_\_\_

.04 = \_\_\_\_\_

## Comparison of Results

16. Compare the results obtained in the **.f06** file with the results on the following page:

POINT-ID = 23

D I S P L A C E M E N T   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	.0	.0	.0	.0
1.999998E-02	G	.0	.0	2.511800E-06	.0	.0	.0
3.999996E-02	G	.0	.0	2.523053E-06	.0	.0	.0

POINT-ID = 33

D I S P L A C E M E N T   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	.0	.0	.0	.0
1.999998E-02	G	.0	.0	4.559710E-06	7.870200E-20	-5.699044E-07	.0
3.999996E-02	G	.0	.0	7.363514E-07	-8.475326E-19	4.972196E-07	.0

POINT-ID = 23

V E L O C I T Y   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	1.029922E-05	.0	.0	.0
1.999998E-02	G	.0	.0	4.035838E-07	.0	.0	.0
3.999996E-02	G	.0	.0	4.712748E-07	.0	.0	.0

POINT-ID = 33

V E L O C I T Y   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	-4.379307E-07	1.492096E-20	-1.481305E-07	.0
1.999998E-02	G	.0	.0	1.220225E-03	-2.594927E-15	-3.394969E-04	.0
3.999996E-02	G	.0	.0	6.578242E-04	3.797481E-15	-1.829342E-04	.0

POINT-ID = 23

A C C E L E R A T I O N   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	1.029922E-01	.0	.0	.0
1.999998E-02	G	.0	.0	1.615566E-04	.0	.0	.0
3.999996E-02	G	.0	.0	-1.353748E-04	.0	.0	.0

POINT-ID = 33

A C C E L E R A T I O N   V E C T O R

TIME	TYPE	T1	T2	T3	R1	R2	
.0	G	.0	.0	-4.379307E-03	1.492096E-16	-1.481305E-03	.0
1.999998E-02	G	.0	.0	-1.464230E+00	-1.464654E-11	4.074106E-01	.0
3.999996E-02	G	.0	.0	1.231636E+00	1.381082E-11	-3.427888E-01	.0

17. MSC/NASTRAN Users have finished this exercise. MSC/PATRAN Users should proceed to the next step.

18. Proceed with the Reverse Translation process, that is importing the **prob7.op2** results file into MSC/PATRAN. To do this, return to the Analysis form and proceed as follows:

◆ **Analysis**

<i>Action:</i>	<b>Read Output2</b>
<i>Object:</i>	<b>Result Entities</b>
<i>Method:</i>	<b>Translate</b>
<b>Select Results File...</b>	
<i>Select Results File</i>	<b>prob7.op2</b>
<b>OK</b>	
<b>Apply</b>	

When the translation is complete bring up the form.

◆ **Results**

<i>Form Type:</i>	<b>Advanced</b>
<i>Select Result Cases</i> <i>(Highlight all cases.)</i>	
<b>Get Results</b>	
<i>Select Result</i>	<b>1.1 Displacements, Translational</b>
<i>Plot Type</i>	<b>XY Plot</b>
<b>Plot Type Options ...</b>	
<b>Global Var...</b>	
<i>Global Variables:</i>	<b>1. Time</b>
<b>Apply</b>	
<b>Result (Y)...</b>	
<i>Results</i>	<b>1.1 Displacements, Translational</b>
<i>Vector Component</i>	<input type="checkbox"/> X <input type="checkbox"/> Y <input checked="" type="checkbox"/> Z
<b>OK</b>	

---

Node IDs:

Node 23

Apply...

New Title or Title Filter:

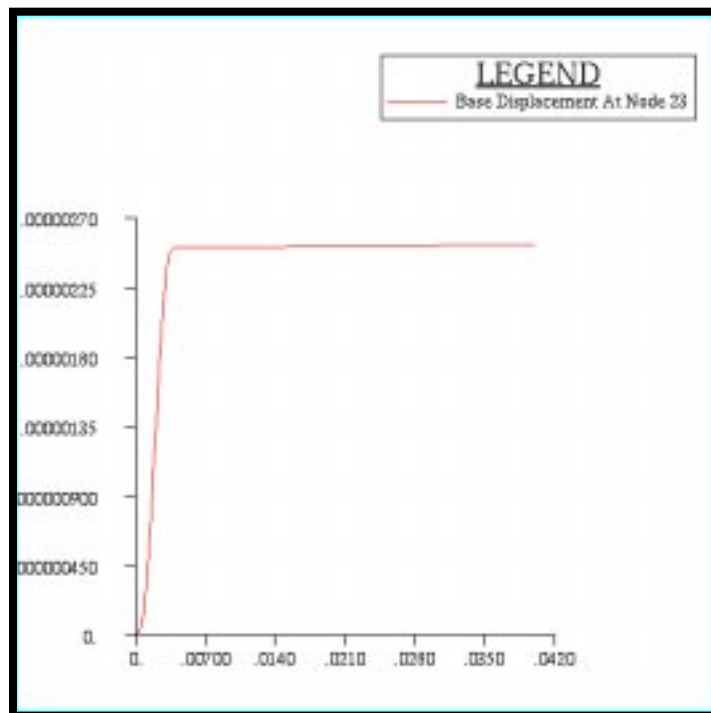
Base Displacement At  
Node 23

Rename

Apply

The output should look similar to Figure 7.3

**Figure 7.3-**Base Displacement at Node 23



19. Repeat the procedure to find the nodal displacement for Node 33.

◆ **Results**

Form Type:

Advanced

Select Result Cases  
(Highlight all cases.)

Get Results

Select Result

1.1 Displacements, Translational

Plot Type

XY Plot



**Plot Type Options ...**

**Global Var...**

*Global Variables:*

**Apply**

**Result (Y)...**

*Results*

*Vector Component*

**OK**

*Node IDs:*

**Apply...**

*Result XY Window Name:*

*New Title or Title Filter:*

**Rename**

**Apply**

**1. Time**

**1.1 Displacements, Translational**

X    Y    Z

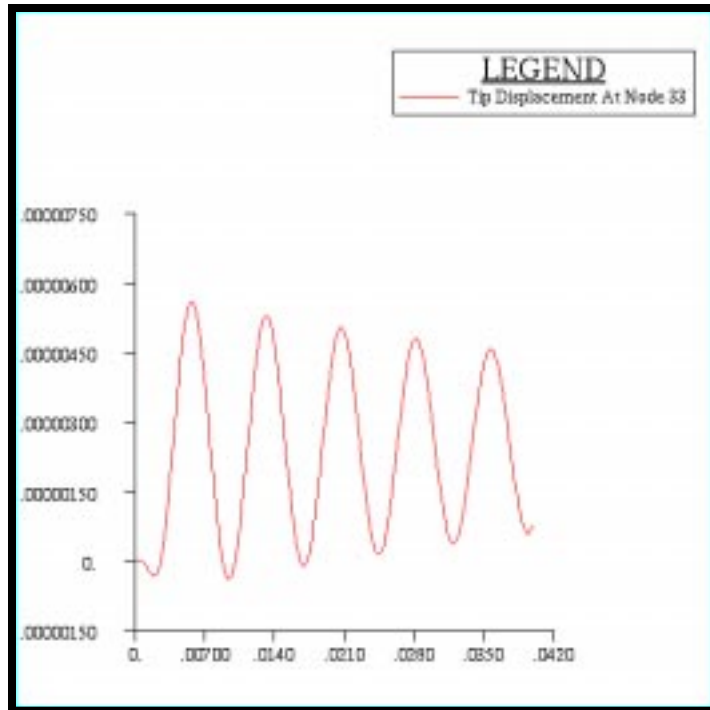
**Node 33**

**XYWindow2**

**Tip Displacement at Node 33**

The output should look similar to Figure 7.4

Figure 7.4- Tip Displacement at Node 33



20. Repeat the procedure to find the nodal velocity for Node 23.

◆ **Results**

Form Type:

**Advanced**

Select Result Cases  
(Highlight all cases.)

**Get Results**

Select Result

**1.1 Velocity, Translational**

Plot Type

**XY Plot**

**Plot Type Options ...**

**Global Var...**

Global Variables:

**1. Time**

**Apply**

**Result (Y)...**

Results

**1.1 Velocity, Translational**

Vector Component

X  Y  Z

**OK**

*Node IDs:*

**Node 23**

**Apply...**

*Result XY Window Name:*

**XYWindow3**

*New Title or Title Filter:*

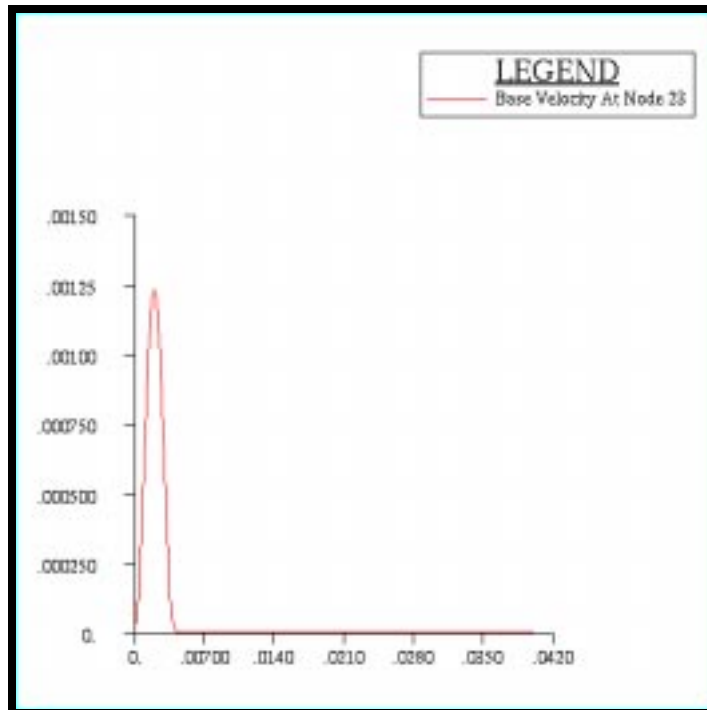
**Base Velocity at Node 23**

**Rename**

**Apply**

The output should look similar to Figure 7.5

**Figure 7.5-Base Velocity at Node 23**



21. Repeat the procedure to find the velocity at Node 33.

◆ **Results**

*Form Type:*

**Advanced**

*Select Result Cases  
(Highlight all cases.)*

**Get Results**

---

*Select Result*

**1.1 Velocity, Translational**

*Plot Type*

**XY Plot**

**Plot Type Options ...**

**Global Var...**

*Global Variables:*

**1. Time**

**Apply**

**Result (Y)...**

*Results*

**1.1 Velocity, Translational**

*Vector Component*

X    Y    Z

**OK**

*Node IDs:*

**Node 33**

**Apply...**

*Result XY Window Name:*

**XYWindow4**

*New Title or Title Filter:*

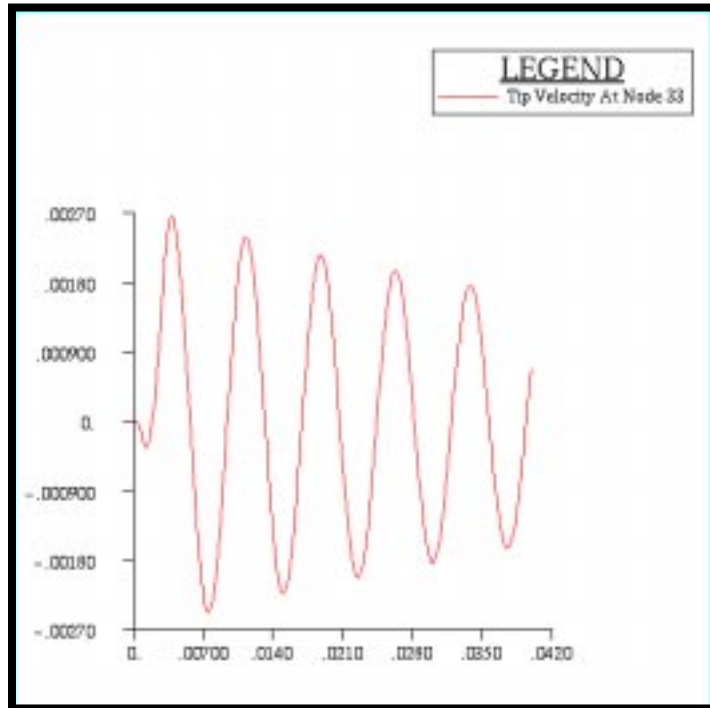
**Tip Velocity at Node 33**

**Rename**

**Apply**

The output should look similar to Figure 7.6

Figure 7.6- Tip Velocity at Node 33



22. Repeat the procedure to find the nodal acceleration for Node 23.

◆ **Results**

*Form Type:*

**Advanced**

*Select Result Cases*  
(Highlight all cases.)

**Get Results**

*Select Result*

**1.1 Acceleration, Translational**

*Plot Type*

**XY Plot**

**Plot Type Options ...**

**Global Var...**

*Global Variables:*

**1. Time**

**Apply...**

**Result (Y)...**

**1.1 Acceleration, Translational**

*Results*

*Vector Component*

X  Y  Z

**OK**

*Node IDs:*

**Node 23**

**Apply...**

*Result XY Window Name:*

**XYWindow5**

*New Title or Title Filter:*

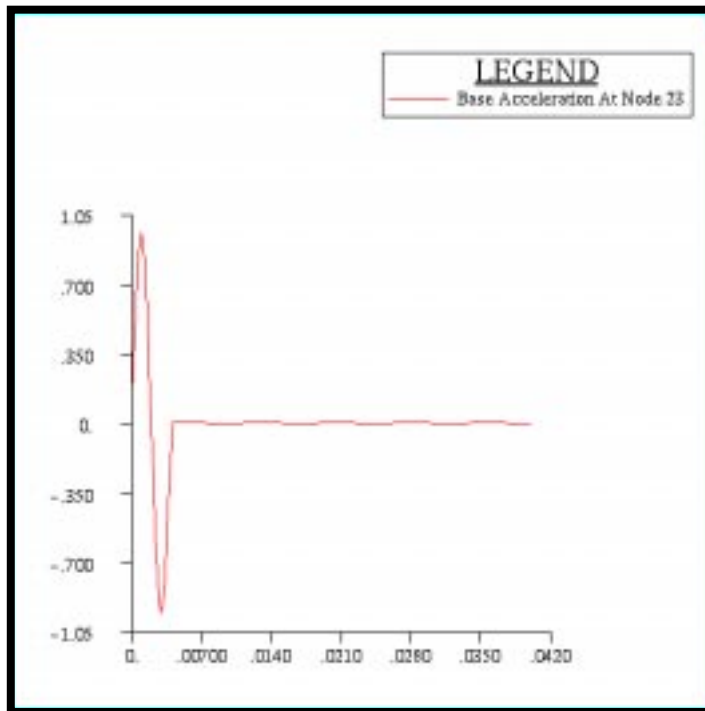
**Base Acceleration at Node 23**

**Rename**

**Apply**

The output should be similar to Figure 7.7

**Figure 7.7-Base Acceleration at Node 23**



23. Repeat the procedure to find the acceleration at Node 33.

◆ **Results**

*Form Type:*

**Advanced**

*Select Result Cases*

*(Highlight all cases.)*

**Get Results**

*Select Result*

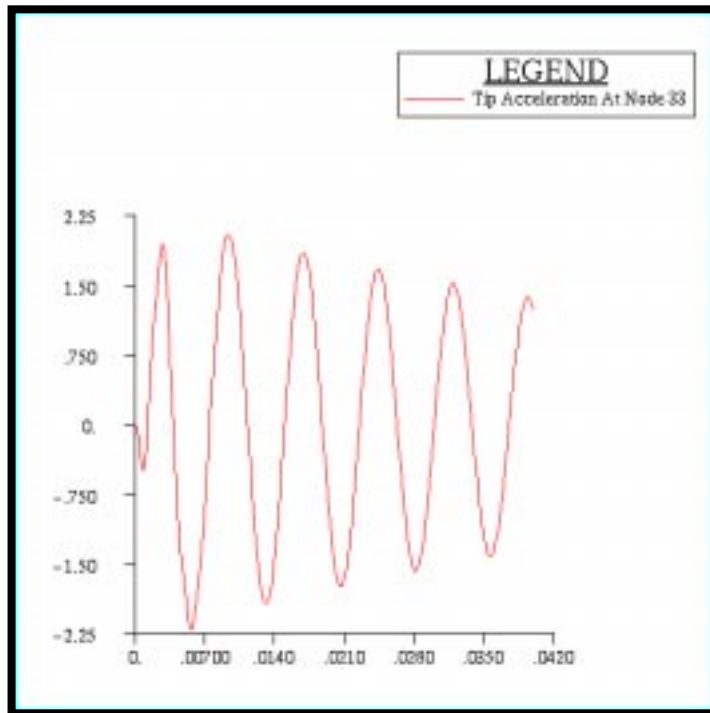
**1.1 Acceleration, Translational**

<i>Plot Type</i>	<input type="text" value="XY plot"/>
<input type="text" value="Plot Type Option ..."/>	
<input type="text" value="Global Var..."/>	
<i>Global Variables:</i>	<input type="text" value="1. Time"/>
<input type="text" value="Apply"/>	
<input type="text" value="Result (Y)..."/>	
<i>Results:</i>	<input type="text" value="1.1 Acceleration, Translational"/>
<i>Vector Component</i>	<input type="checkbox"/> X <input type="checkbox"/> Y <input checked="" type="checkbox"/> Z
<input type="text" value="OK"/>	
<i>Node IDs:</i>	<input type="text" value="Node 33"/>
<input type="text" value="Apply..."/>	
<i>Result XY Window Name:</i>	<input type="text" value="XYWindow6"/>
<i>New Title or Title Filter:</i>	<input type="text" value="Tip Acceleration at Node 33"/>
<input type="text" value="Rename"/>	
<input type="text" value="Apply"/>	

The output should look similar to Figure 7.8

---

**Figure 7.8- Tip Acceleration at Node 33**



Quit MSC/PATRAN when you are finished with this exercise.